

## Self-Triggered Formation Control of an Asynchronous Multiple Underwater Unmanned Vehicle Under Communication Delay

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To address the problem of multiple underwater unmanned vehicle (UUV) formation tracking under bounded communication delay, considering the existence of model parameter perturbations of UUVs as well as external ocean current disturbances, a distributed robust model predictive controller (DRMPC) with a dual closed-loop control structure and self-triggering mechanism is proposed. Among them, the outer-loop controller transforms the target trajectory into the target speed required by the inner-loop controller, and the inner-loop controller solves the effects of model parameter perturbations and bounded disturbances by RMPC and a finite-time extended state observer (FESO). To reduce the communication load of the system, a self-triggering mechanism is designed to determine the next triggering moment while solving the outer-loop optimization problem. The feasibility and stability of this controller are analyzed via mathematical induction and the ISpS Lyapunov theory. Finally, the effectiveness of this formation controller is validated through simulation experiments with multi-UUV.

**Keywords:** Underwater unmanned vehicle; formation trajectory tracking; asynchronous communication; self-triggering mechanism; distributed robust model predictive control.

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### 1. Introduction

Since entering the 21st century, technological development has accelerated the consumption of energy, fossil energy sources are becoming increasingly depleted, and countries around the world are competing to begin to recognize and explore the ocean [1, 2]. However, complex marine environments, such as those characterized by communication difficulties, dimness, high pressure, and corrosiveness, hinder human exploration of the ocean, and UUV technology has become an important means for ocean exploration because of its unmanned, high maneuverability, mass deployment [3, 4], etc. As a new type of carrier in ocean engineering, UUV technology has important strategic

significance in the development of marine resources and enhancement of marine military power [5, 6].

With the deepening of ocean development layer by layer, single UUV is difficult to meet the demand of executing complex and diversified missions with wide coverage due to its limited functional load and control range capability, prompting the transition of UUV application research towards multi-UUV formation control [7, 8]. Compared with a single UUV, UUV formation has stronger robustness, greater communication capability, greater flexibility, higher operational efficiency, and wider operational range [9]. Multi-UUV formation technology improves the execution efficiency in the fields of underwater search and rescue, environmental detection, surveillance and reconnaissance, and gaming and roundup [10]. At the same time, it can perform complex tasks that cannot be accomplished by traditional manned platforms, such as all-around, long-cycle, unmanned, and systematic defense alerts offshore and target sea area cleaning and reconnaissance. In complex

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marine environments, model parameter perturbations and external disturbances are huge obstacles to formation control [11–13]. At the same time, since the main means of underwater communication for multi-UUV formation is hydroacoustic communication, it has weak communication characteristics, such as a large propagation delay and susceptibility to communication interruption. The study of formation control methods applicable to weak communication conditions is important for enhancing the reliability, redundancy, and robustness of the formation system.

As a current research hotspot, UUV technology has been widely researched by researchers from various countries, from mission planning to platform control, but there are tricky challenges in multi-UUV formation control. Currently, frequent formation control algorithms include the leader-following method [14], artificial potential field method [15], reinforcement learning [16], and virtual structure method [17]. Multi-UUV formation requires collaborative execution of ocean missions according to the expected formation. Reliable communication between UUVs is crucial for the stable control of formation systems, and sampling mechanisms can reduce the communication frequency and better meet the needs of the formation system. The sampling methods applied in multi-agent systems can be divided into synchronous sampling and asynchronous sampling. In marine environments, due to limited technology, the synchronization of samplers cannot be achieved between UUVs, increasing the practicality of asynchronous sampling. Ding and Zheng [18] proposed an asynchronous sampling tracking protocol that guarantees bounded tracking error via the Lyapunov-Krasovskii method and designed a consistent controller gain matrix to achieve the formation tracking of heterogeneous nonlinear multi-agent system. For the networked mobile robot formation problem with communication delays, Peng *et al.* [19] solved the asynchronous clock formation problem by detecting the timing of synchronization points and combining the temporal and sequential characteristics of the messages to ensure that the messages have the correct order at the receiver. Han and Zeng [20] designed an asynchronous pulse mechanism that solves asymmetric feedback saturation by the convex hull method to enable the precise convergence of the system with communication delays. To improve control performance and decrease the energy consumption of a second-order multi-agent system, Yang *et al.* [51] proposed a distributed asynchronous control protocol, which reduces the interaction frequency and improves the formation consistency. Zhang and Su designed a distributed discrete observer using relative sampling output information and solved the asynchronous consensus problem for a multi-intelligent body system under switching topology by asynchronous sampling data distributed output feedback control [47]. Asynchronous sampling

improves the control efficiency, but it takes up more computational resources and increases the number of communications between intelligences.

In practical applications, unnecessary information interactions occur when the formation system is close to stabilization, leading to energy waste in the formation system. Significantly reducing communication and computational energy consumption while ensuring the formation control performance of multi-intelligent body systems is the key to improving system stability. The event-triggered method is an important technique for reducing control energy consumption in intelligent agent cooperative control, controlling the necessary interactions and computations of members in the formation [21]. To address the problem of multi-surface ship formation under the influence of model parameter perturbation and current disturbance, Zheng *et al.* [13] designed a dynamic event triggering mechanism, and proposed a model parameter-free formation controller by combining sliding mode control and adaptive control. Song *et al.* [22] introduced internal dynamic variables into the trigger function, and combined the adaptive method and the minimum learning parameter method to design a multi-unmanned vessel formation controller based on edge event triggering under the influence of current disturbances. For the formation control problem of fractional-order multi-agent system, Meng *et al.* [23] proposed a dynamic event triggering SMC controller, designed a dynamic triggering mechanism using combined measurement vectors, and dynamically adjusted the threshold parameters of the triggering condition through auxiliary variables. Xia *et al.* [24] proposed an activatable dynamic event-triggered protocol, designed an event-triggered detector and a controller by combining adaptive control, and demonstrated the effectiveness of this approach with multi-spacecraft formation control. Yu *et al.* [25] proposed an event triggering mechanism based on position and direction and designed a high-gain extended state observer combined with a backstepping method, which reduces the requirement for velocity information and realizes the distributed formation control of multi-ground vehicles. Ma *et al.* [48] designed an event-triggered fault-tolerant formation controller based on the framework of the backtracking method in finite time to realize the formation control of multiple UAVs under external perturbations and flight faults.

Although the event-triggered mechanism effectively reduces the communication frequency, it requires real-time measurement of the system state. The continuous computation of the event-triggered mechanism will cause excessive resource loss, in order to solve this problem [26], the self-triggered control is proposed, which predicts the next triggering moment based on the information of the system at the current triggering moment [27]. For the second-order

multi-agent system affine formation problem, Ma *et al.* [28] designed an event-triggered proportional-integral controller based on sampled data and combined with the self-triggered control, which effectively reduces the frequency of information interactions and controller computations. Zhao *et al.* [29] designed an interval observer and an event-triggered state compensator, and obtained the local triggering time through a self-triggering controller to achieve formation containment control of heterogeneous multi-agent system with external perturbations and missing state quantities [30]. For the problem of multi-spacecraft attitude synchronization with time-varying communication delay, Xie *et al.* [31] established a formation attitude dynamics model, proposed a dynamic event formation control strategy combined with a nonlinear observer, and designed a self-triggering function based on the triggering information, which avoids the continuous calculation of event triggering. Chai *et al.* [32] designed a self-triggering control scheme based on an observer, which reduces communication energy consumption while ensuring formation error and state estimation error, and achieves formation control of multi-intelligent body without directional topology. Jin *et al.* [33] designed a dual closed-loop self-triggering control framework considering unknown wind perturbation, which realized stable flight of a fixed-wing UAV formation under speed and overload constraints, while avoiding continuous rapid sampling. Zhao *et al.* [49] designed a self-triggered approximate optimal neural control method for nonlinear systems based on ADP, which obtains the update time of the next control strategy through the self-triggering mechanism and avoids the continuous monitoring of the system state by hardware devices.

Distributed model predictive control (DMPC) deals with multi-constraint control problems through rolling optimization and feedback correction, and achieves accurate control through prediction [34, 35]. With the development of DMPC in multi-agent formation control, triggered MPC enters into the limelight. To address the consistency problem of linear discrete MAS, Zhan *et al.* [36] weighed the control inputs and triggering intervals and designed a distributed self-triggered MPC method to realize the dynamic consensus of multi-agents. Mi *et al.* [37] designed a DMPC cooperative controller for dynamically decoupled agent systems based on a self-triggering mechanism, which guarantees the control performance and reduces the communication load at the same time. For multi-constrained linear time-invariant systems with disturbances, Kolarijani *et al.* [38] proposed a sampling strategy based on set theory and designed a robust formation MPC control algorithm to achieve constraint tightening. Zou *et al.* [39] designed a dynamic triggering mechanism based on the information of neighboring nodes, which balances resource consumption and control performance, and realizes asynchronous

coordinated predictive control of multi-agent under perturbations. Wei *et al.* [40] obtained the next triggering time by solving local optimization problems and proposed a self-triggering min-max DMPC control algorithm to achieve nonlinear MAS formation control under communication delay and disturbance influence. Yang *et al.* [50] constructed a robust constraint set based on the estimation information of a discrete nonlinear perturbation observer and designed a self-triggered distributed MPC controller combined with an adaptive prediction mechanism to realize the cooperative control of a nonlinear system under perturbation and systematic constraints. So far, DMPC has not been applied to multi-UUV formation control with communication delays, and there are two shortcomings in related studies on triggered DMPC: one is that the control object has a simple nonlinear model or only kinematic models have been used; the other is that the effects of communication delays, time-varying disturbances, and model parameter perturbations on the formation control were not simultaneously considered.

To solve the aforementioned issues and enable the triggered DMPC to be applied to the multi-UUV formation control with multiple constraints, nonlinearity and strong coupling in complex environments, as well as to improve the efficiency of solving the communication delay problem, this paper proposes a DRMPC formation controller based on a dual closed-loop structure and self-triggering mechanism, which reduces information redundancy while enabling the multi-UUV system to maintain the target formation and accurately track the target trajectory under model parameter perturbations and ocean current disturbances. The main contributions of the paper are summarized as follows:

- (1) A self-triggered DRMPC formation controller is proposed for the first time to address the multi-UUV 3D formation problem under bounded time-varying communication delay and multiple constraints. This controller realizes asynchronous communication among formation members, reduces the communication load, and reduces the information redundancy.
- (2) The RMPC and FESO in the inner-loop controller work together to solve the problem of bounded disturbances and model parameter perturbations affecting the formation control and improve the robustness of the system.
- (3) The stability and feasibility of the self-triggered formation controller at the triggering moment are analyzed by using the input-to-state practical stable (ISpS) theorem and mathematical induction, which provides a new idea for DMPC to handle multi-constraint, nonlinear multi-agent collaborative control under time-varying communication delays.

The remainder of the paper is organized as follows: Section 2 establishes the UUV mathematical model and provides important lemmas and problem description. Section 3 provides a detailed introduction to the design and theoretical analysis of a self-triggered formation controller. Section 4 verifies the effectiveness of the self-triggered formation controller through simulation experiments. Section 5 briefly summarizes the paper and suggests future research directions.

Notations:  $\mathbb{N}_{[m,n]}$  denotes a positive integer within  $[m, n]$ ,  $I_n \in \mathbb{R}^{n \times n}$  denotes the identity matrix,  $\mathbf{0}_{n \times m}$  denotes the  $n \times m$ -dimensional zero matrix, and  $c(\cdot)$ ,  $s(\cdot)$  and  $t(\cdot)$  denote the cosine, sine and tangent functions, respectively. assuming  $v \in \mathbb{R}^n$ ,  $\mu^\alpha(v) = \text{sign}(v)|v|^\alpha$ , and  $\text{sign}(\cdot)$  denotes the sign function.

## 2. Preliminaries

### 2.1. UUV mathematical model

The UUV studied in the paper is a three-dimensional flat structure that can be approximated as left-right symmetric with large damping in the transverse roll direction; therefore, motion in the transverse roll direction is usually ignored when designing controllers [41]. To study the motion law of UUV, we selected the inertial coordinate system  $O_E X_E Y_E Z_E$  and the body-fixed coordinate system  $O_B X_B Y_B Z_B$  to describe the motion of UUV, and the schematic diagram of the coordinate systems is shown in Fig. 1. Usually,  $\eta = [x; y; z; \theta; \psi]$  is used to describe the position and attitude of the UUV in the inertial coordinate system, and  $\nu = [u; v; w; q; r]$  is used to describe the velocity and angular velocity of the UUV in the body-fixed coordinate system. The UUV is controlled by the actuator that generates force and torque to control its motion, and the control input vector is  $\tau = [\tau_u; \tau_v; \tau_w; \tau_q; \tau_r]$ .

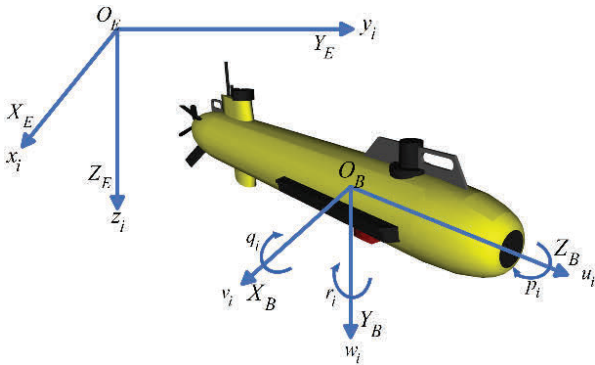


Fig. 1. UUV coordinate system.

Taking  $UUV_i$  as an example, its system model is Eq. (1):

$$\begin{aligned} \dot{\eta}_i &= J_i(\eta_i) \nu_i = f_1(\eta_i, \nu_i) \\ \dot{\nu}_i &= M_i^{-1} [\tau_i - C_i(\nu_i) \nu_i - D_i(\nu_i) \nu_i - g_i(\eta_i)] = f_2(\nu_i, \tau_i), \end{aligned} \quad (1)$$

where  $M_i$  and  $C_i(\nu_i)$  are the inertia and Coriolis centripetal matrices containing the hydrodynamic added mass, respectively, and  $D_i(\nu_i)$  is the hydrodynamic damping coefficient matrix.  $g_i(\eta_i)$  is the restoring force and moment, which is assumed to cancel each other since gravity and buoyancy are acting on the same line, and hence  $g_i(\eta_i) = \mathbf{0}$ .  $J_i(\eta_i)$  is the rotation matrix, i.e.,

$$J_i(\eta_i) = \begin{bmatrix} c(\theta)c(\psi) & -s(\psi) & s(\theta)c(\psi) & \\ c(\theta)s(\psi) & c(\psi) & s(\theta)s(\psi) & \mathbf{0}_{3 \times 2} \\ -s(\theta) & 0 & c(\theta) & \\ & \mathbf{0}_{2 \times 3} & 1 & 0 \\ & & 0 & 1/c(\theta) \end{bmatrix}. \quad (2)$$

### 2.2. Problem formulation

The information interaction between UUVs is represented by a directed graph  $G = (\mathcal{V}, \mathcal{E}, \mathcal{W})$  containing  $n$  nodes, and each UUV in the formation corresponds to a node in the directed graph  $G$ , where  $\mathcal{V} = \{1, 2, \dots, n\}$ ,  $\mathcal{E} = \{\varepsilon_{ij} : i, j \in \mathcal{V}, i \neq j\} \subseteq \mathcal{V} \times \mathcal{V}$ , and  $\mathcal{W} = [w_{ij}] \in \mathbb{R}^{n \times n}$  are the vertex set, edge set, and adjacency matrix of the directed graph  $G$ , respectively. The set of adjacent individuals of  $UUV_i$  is  $\mathcal{N}_i = \{j : \varepsilon_{ji} \in \mathcal{E}\}$ , and the cardinality of set  $\mathcal{N}_i$  is  $n_i$ . In the adjacency matrix, when  $\varepsilon_{ji} \in \mathcal{E}$ ,  $w_{ij} = 1$ , i.e.,  $UUV_i$  can receive the information sent by  $UUV_j$ ; otherwise,  $w_{ij} = 0$ , which means that  $UUV_i$  cannot receive the information from  $UUV_j$  [42].

Under the directed graph, this paper investigates the multi-UUV formation tracking problem, aiming to design an asynchronous self-triggered controller so that the positions and velocities of all UUVs reach a relatively stable state under the influence of communication delays, bounded disturbances, and model parameter perturbations. The formation control objective is defined as follows:

$$\begin{aligned} \lim_{t_k^i \rightarrow \infty} \{\eta_{i0}(t_k^i)\} &= \lim_{t_k^i \rightarrow \infty} \{\eta_i(t_k^i) - \eta_0(t_k^i) - \Delta_i\} \\ &= 0, \quad \forall i \in \mathbb{N}_{[1,n]}, \end{aligned} \quad (3)$$

$$\begin{aligned} \lim_{t_k^i \rightarrow \infty} \{\eta_{ij}(t_k^i)\} &= \lim_{t_k^i \rightarrow \infty} \{\eta_i(t_k^i) - \eta_j(t_k^i) - \Delta_{ij}\} \\ &= 0, \quad \forall i, j \in \mathbb{N}_{[1,n]}, \end{aligned} \quad (4)$$

$$\begin{aligned} \lim_{t_k^i \rightarrow \infty} \{\nu_{ij}(t_k^i)\} &= \lim_{t_k^i \rightarrow \infty} \{\nu_i(t_k^i) - \nu_j(t_k^i)\} \\ &= 0, \quad \forall i, j \in \mathbb{N}_{[1,n]}, \end{aligned} \quad (5)$$

where  $\eta_i(t_k^i)$  and  $\nu_i(t_k^i)$  are the position and velocity of  $UUV_i$  at moment  $t_k^i$ , respectively;  $\eta_0(t_k^i)$  is the position of virtual

leader  $UUV_0$  at moment  $t_k^i$ ;  $\Delta_i$  is the target distance between  $UUV_i$  and  $UUV_0$ ; and  $\Delta_{ij}$  is the target distance between  $UUV_i$  and  $UUV_j$ .

### 2.3. Important lemmas

**Lemma 1 ([40]).** For a system  $x(t_k^i + 1) = f(x(t_k^i), u(t_k^i))$  with a robust invariant set, there exists a positive definite function  $V(\cdot) : \mathbb{R}^n \rightarrow \mathbb{R}$  satisfying  $\alpha_1(\|x\|) \leq V(x) \leq \alpha_2(\|x\|) + c_1$  and  $V(f(x, u)) - V(x) \leq -\alpha_3(\|x\|) + \gamma(\|w\|) + c_2$ , where  $c_1$  and  $c_2$  are positive real numbers,  $\alpha_1(\cdot)$ ,  $\alpha_2(\cdot)$ , and  $\alpha_3(\cdot)$  are  $\mathcal{K}_\infty$  functions, and  $\gamma(\cdot)$  is an  $\mathcal{K}$  function; then, the system is input-to-state practically stable (ISpS) in the robust invariant set.

**Lemma 2 ([43]).** For a perturbed system  $x(t_k^i + 1) = f(x(t_k^i), d(t_k^i))$ , if  $f(x(t_k^i), d(t_k^i)) \in \Omega$ , for any  $x \in \Omega$  and  $d(t_k^i) \in \Xi$ , then the set  $\Omega$  is the robust invariant set of the system.

**Lemma 3 ([43]).** For a discrete system  $x(t_k^i + 1) = f(x(t_k^i), u(t_k^i), d(t_k^i))$ ,  $\Xi_N(\Omega)$  is said to be a robust  $N$ -step set of sets  $\Omega$  if for any  $x(t_k^i) \in \Xi_N(\Omega) \subset \mathbb{R}^n$  and  $d(t_k^i) \in \mathbb{W}$ , there exists a feasible sequence of control inputs  $U_N = \{u(0|t_k^i), \dots, u(N-1|t_k^i)\}$  such that  $x(N|t_k^i) \in \Omega$ , subject to control input and state constraints, i.e.  $\Xi_N(\Omega) = \{x(t_k^i) \in \Omega | \exists U_N : x(N|t_k^i) \in \Omega, \forall d(t_k^i) \in \mathbb{W}\}$ .

## 3. Controller Design

In this section, an asynchronous communication strategy with communication delay is proposed, and an extended

state observer and RMPC are designed to resist the effects of model parameter perturbations and current disturbances. Then, a self-triggered distributed formation controller with a dual closed-loop structure is proposed, and the control structure diagram is shown in Fig. 2. Finally, the formation controller is theoretically analyzed.

### 3.1. Asynchronous communication strategy

First, we design the asynchronous communication strategy under communication delay, and Fig. 3 is a schematic diagram of asynchronous communication.  $t_k^i$  is the triggering moment of intermittent communication of  $UUV_i$ , at which  $UUV_i$  receives the prediction information from neighboring UAVs, and calculates the optimal communication interval and the new control inputs.  $\tau_k^{ij}$  is the communication delay when  $UUV_j$  sends information to  $UUV_i$ . If  $\exists i, j \in \mathcal{N}, i \neq j$ , and  $t_k^i, t_k^j$  are not equal and independent of each other, the formation system communicates asynchronously, and the UAVs cannot update their states simultaneously.

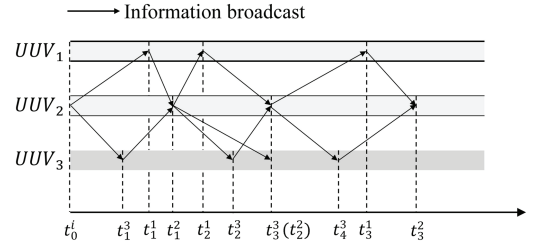


Fig. 3. Asynchronous communication schematic.

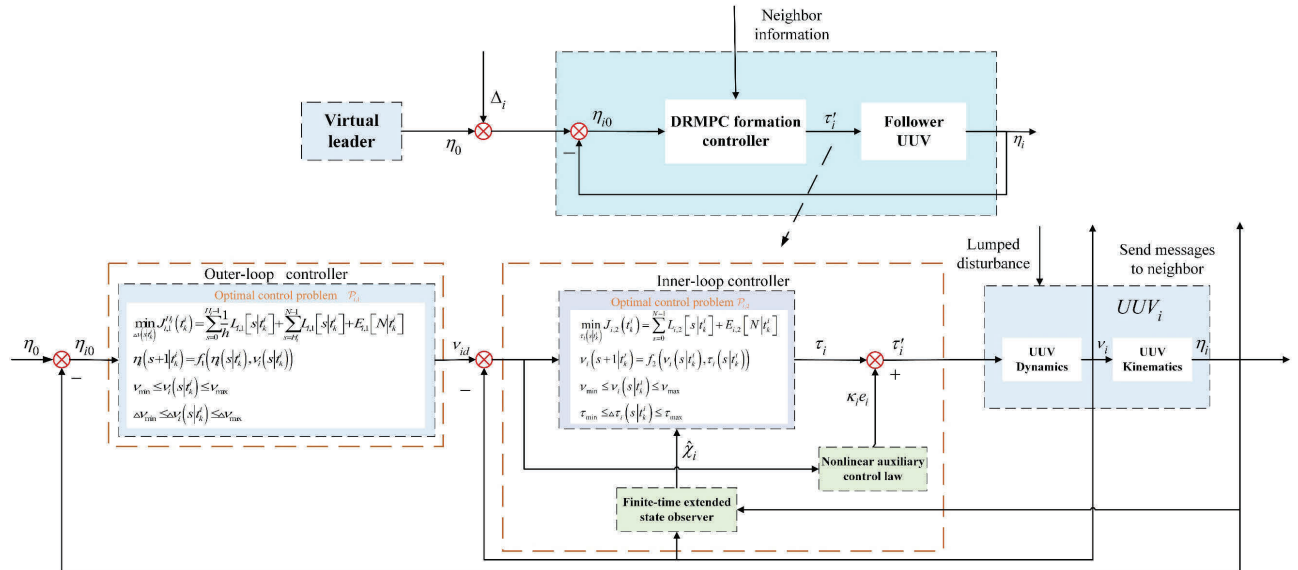


Fig. 2. Block diagram of the formation controller.

**Assumption 1.** When the control interval and the communication delay are large, UUVs cannot receive new information for a long time, thereby causing instability in multi-UUV formation systems. Taking  $UUV_i$  as an example, the local sampling moment  $t_k^i$  and the communication delay  $\tau_k^{ij}$  need to satisfy the following conditions[40]: (1)  $T \leq t_{k+1}^i - t_k^i \leq H_{\max} \cdot T$ ; (2)  $0 \leq \tau_k^{ij} \leq T_{\max} \cdot T$ , where  $H_{\max}$  and  $T_{\max}$  are both integer multiples of the control period, which denote the maximal sampling interval and the maximal communication delay, respectively.

At moment  $t_k^i$ , the predicted state sequence broadcasted by  $UUV_i$  is  $\eta_i^b(t_k^i) = (\eta_i^b(0|t_k^i), \dots, \eta_i^b(N-1|t_k^i))$ , and its expression is:

$$\eta_i^b(s|t_k^i) = \begin{cases} \eta_i^*(s|t_k^i), & s \in \mathbb{N}_{[0, N-1]} \\ 0, & s \in \mathbb{N}_{[N, \infty)} \end{cases}, \quad (6)$$

where  $\eta_i^*(s|t_k^i)$  denotes the optimal prediction state. The predicted velocity sequence for  $UUV_i$  broadcasting is  $\nu_i^b(t_k^i) = (\nu_i^b(0|t_k^i), \dots, \nu_i^b(N-1|t_k^i))$ , and its expression is:

$$\nu_i^b(s|t_k^i) = \begin{cases} \nu_i^*(s|t_k^i), & s \in \mathbb{N}_{[0, N-1]} \\ 0, & s \in \mathbb{N}_{[N, \infty)} \end{cases}, \quad (7)$$

where  $\nu_i^*(s|t_k^i)$  denotes the optimal prediction velocity. Assuming that  $\eta_j^b(t_k^{ij})$  and  $\nu_j^b(t_k^{ij})$  are the latest state prediction information and velocity prediction information sent by  $UUV_j$  to  $UUV_i$  at moment  $t_k^{ij}$ , respectively,  $t_k^{ij} = \max\{t_l^j \in \mathbb{N} | t_l^j < t_k^i\}$ , we can categorize them into the following two scenarios according to the different communication delays:

- (1) When  $0 \leq \tau_k^{ij} \leq t_k^i - t_k^{ij}$ ,  $UUV_i$  can receive the latest information from  $UUV_j$  at moment  $t_k^i$ .
- (2) When  $t_k^i - t_k^{ij} < \tau_k^{ij} \leq T_{\max} \cdot T$ ,  $UUV_i$  cannot receive the latest information from  $UUV_j$  at moment  $t_k^i$ , so  $UUV_i$  can only use the information  $\eta_j^b(t_k^{ij})$  and  $\nu_j^b(t_k^{ij})$  received at the previous moment.

Thus, the predictive information  $\eta_j(t_k^i)$  and  $\nu_j(t_k^i)$  of the neighboring  $UUV_j$  used by  $UUV_i$  in the subsequent local optimization problem can be constructed as

$$\begin{aligned} \eta_j(t_k^i) &= (\eta_j^b(0|t_k^i), \eta_j^b(1|t_k^i), \dots, \eta_j^b(N-1|t_k^i)) \\ \nu_j(t_k^i) &= (\nu_j^b(0|t_k^i), \nu_j^b(1|t_k^i), \dots, \nu_j^b(N-1|t_k^i)), \end{aligned} \quad (8)$$

$$\eta_j^b(s|t_k^i) = \begin{cases} \eta_j^*(s|t_k^i), & s \in \mathbb{N}_{[0, N-1]} \\ 0, & s \in \mathbb{N}_{[N, H+N+T_{\max}]} \end{cases}, \quad (9)$$

$$\nu_j^b(s|t_k^i) = \begin{cases} \nu_j^*(s|t_k^i), & s \in \mathbb{N}_{[0, N-1]} \\ 0, & s \in \mathbb{N}_{[N, H+N+T_{\max}]} \end{cases}. \quad (10)$$

### 3.2. FESO design

First, we give an auxiliary variable  $\omega_i = J_i \nu_i$  and transform the model (1) to obtain a new UUV model:

$$\begin{aligned} \dot{\eta}_i &= \omega_i \\ \dot{\omega}_i &= J_i M_i^{-1} \tau_i - J_i M_i^{-1} J_i h(\omega_i) \omega_i + \mathcal{X}_i. \end{aligned} \quad (11)$$

**Assumption 2.** Define the function  $h(\omega)\omega = [C(\omega) + D(\omega)]\omega$ . Since the actuator of the UUV has certain physical constraints that limit the motion of the UUV, there exists a coefficient  $\mu$  such that  $\|h(\omega)\omega\| \leq \mu\|\omega\|^2$ [41].

where  $\mathcal{X}_i = J_i S_i \nu_i + \delta_i$  is the set total perturbation,  $\delta_i$  denotes the model parameter perturbation, and the matrix  $S_i$  is:

$$S_i = \begin{bmatrix} 0 & -r_i & q_i & \\ r_i & 0 & r_i t(\theta_i) & \mathbf{0}_{3 \times 2} \\ -q_i & -r_i t(\theta_i) & 0 & \\ & \mathbf{0}_{2 \times 3} & 0 & 0 \\ & & 0 & q_i t(\theta_i) \end{bmatrix}. \quad (12)$$

**Assumption 3.** The set total uncertainty  $\mathcal{X}_i$  in Eq. (11) is bounded and continuously differentiable, its derivative is bounded, i.e.,  $\|\dot{\mathcal{X}}_i\| \leq \bar{\mathcal{X}}$ , and  $\bar{\mathcal{X}}$  is a positive constant [44].

In this section, an FESO is designed to estimate the uncertainty of the UUV system at a fixed time.  $\hat{\eta}_i$ ,  $\hat{\omega}_i$ , and  $\hat{\mathcal{X}}_i$  are defined as the estimates of  $\eta_i$ ,  $\omega_i$ , and  $\mathcal{X}_i$ , and the estimation errors are defined as  $\mathbf{e}_{i1} = \hat{\eta}_i - \eta_i$ ,  $\mathbf{e}_{i2} = \hat{\omega}_i - \omega_i$ , and  $\mathbf{e}_{i3} = \hat{\mathcal{X}}_i - \mathcal{X}_i$ . The designed FESO is:

$$\begin{aligned} \dot{\hat{\eta}}_i &= \hat{\omega}_i - b_{i1}(\text{sig}^{a_{i1}}(\mathbf{e}_{i1}) + \mathbf{e}_{i1}) \\ \dot{\hat{\omega}}_i &= \hat{\mathcal{X}}_i - J_i M_i^{-1} h(\hat{\omega}_i) \hat{\omega}_i + J_i M_i^{-1} \tau_i - b_{i2}(\text{sig}^{a_{i2}}(\mathbf{e}_{i2}) + \rho_i \mathbf{e}_{i1}) \\ \dot{\hat{\mathcal{X}}}_i &= -b_{i3}(\text{sig}^{a_{i3}}(\mathbf{e}_{i3}) + \rho_i \mathbf{e}_{i1}) \end{aligned} \quad (13)$$

where  $\rho_i = |\mathbf{e}_{i1}|^{a_{i1}-1}$ , the design parameters of the FESO are  $a_{i1} \in (0.67, 1)$ ,  $a_{i2} = 2a_{i1} - 1$ ,  $a_{i3} = 3a_{i1} - 2$ ,  $b_{i1}$ ,  $b_{i2}$ , and  $b_{i3}$  are positive.

**Theorem 1.** For system (11), if the assumptions hold, the estimation error converges to the origin in finite time, i.e.,  $\hat{\eta}_i \equiv \eta_i$ ,  $\hat{\omega}_i \equiv \omega_i$ , and  $\hat{\mathcal{X}}_i \equiv \mathcal{X}_i$ . Therefore, compensation for model parameter perturbations can be achieved by  $\hat{\mathcal{X}}_i$ .

**Proof.** Derive the estimation error:

$$\begin{aligned}\dot{\mathbf{e}}_{i1} &= \mathbf{e}_{i2} - b_{i1}(\text{sig}^{a_{i1}}(\mathbf{e}_{i1}) + \mathbf{e}_{i1}) \\ \dot{\mathbf{e}}_{i2} &= \mathbf{e}_{i3} - J_i \mathbf{M}_i^{-1} J_i (\mathbf{h}(\hat{\omega}_i) \hat{\omega}_i - \mathbf{h}(\omega_i) \omega_i) - b_{i2}(\text{sig}^{a_{i2}}(\mathbf{e}_{i1}) + \rho_i \mathbf{e}_{i1}) \\ \dot{\mathbf{e}}_{i3} &= -b_{i3}(\text{sig}^{a_{i3}}(\mathbf{e}_{i1}) + \rho_i \mathbf{e}_{i1}) - \dot{\mathcal{X}}_i\end{aligned}\quad (14)$$

We introduce the auxiliary variable  $\Gamma_i = [\text{sig}^{a_{i1}}(\mathbf{e}_{i1}) + \mathbf{e}_{i1}; \mathbf{e}_{i2}; \mathbf{e}_{i3}]$ . Theorem 1 holds if the auxiliary variable  $\Gamma_i$  converges to zero in finite time. We take the derivative of variable  $\Gamma_i$

$$\begin{aligned}\dot{\Gamma}_i &= \begin{bmatrix} a_{i1} |\mathbf{e}_{i1}|^{a_{i1}-1} \dot{\mathbf{e}}_{i1} + \dot{\mathbf{e}}_{i1} \\ \dot{\mathbf{e}}_{i2} \\ \dot{\mathbf{e}}_{i3} \end{bmatrix} \\ &= \begin{bmatrix} a_{i1} |\mathbf{e}_{i1}|^{a_{i1}-1} [\mathbf{e}_{i2} - b_{i1}(\text{sig}^{a_{i1}}(\mathbf{e}_{i1}) + \mathbf{e}_{i1})] \\ 0.5 [\mathbf{e}_{i3} - b_{i2}(\text{sig}^{a_{i2}}(\mathbf{e}_{i1}) + \rho_i \mathbf{e}_{i1})] \\ -0.5 b_{i3}(\text{sig}^{a_{i3}}(\mathbf{e}_{i1}) + \rho_i \mathbf{e}_{i1}) \end{bmatrix} + \begin{bmatrix} \mathbf{0}_{5 \times 1} \\ \mathbf{0}_{5 \times 1} \\ -\dot{\mathcal{X}}_i^T \end{bmatrix} \\ &\quad + \begin{bmatrix} \mathbf{e}_{i2} - b_{i1}(\text{sig}^{a_{i1}}(\mathbf{e}_{i1}) + \mathbf{e}_{i1}) \\ 0.5 [\mathbf{e}_{i3} - b_{i2}(\text{sig}^{a_{i2}}(\mathbf{e}_{i1}) + \rho_i \mathbf{e}_{i1})] \\ -0.5 b_{i3}(\text{sig}^{a_{i3}}(\mathbf{e}_{i1}) + \rho_i \mathbf{e}_{i1}) \end{bmatrix} + \begin{bmatrix} \mathbf{0}_{5 \times 1} \\ \mathbf{V} \mathbf{h}_i^T \\ \mathbf{0}_{5 \times 1} \end{bmatrix} \\ &= \text{diag}\{l_i, l_i, l_i\} \begin{bmatrix} -a_{i1} b_{i1} I_5 & a_{i1} I_5 & 0 \\ -0.5 b_{i2} I_5 & 0 & 0.5 \rho_i^{-1} I_5 \\ -0.5 b_{i3} \rho_i I_5 & 0 & 0 \end{bmatrix} \Gamma_i \\ &\quad + \begin{bmatrix} -b_{i1} I_5 & I_5 & 0 \\ -0.5 b_{i2} \rho_i I_5 & 0 & 0.5 I_5 \\ -0.5 b_{i3} \rho_i I_5 & 0 & 0 \end{bmatrix} \Gamma_i + G_{i3} + G_{i4} \\ &= \text{diag}\{l_i, l_i, l_i\} G_{i1} \Gamma_i + G_{i2} \Gamma_i + G_{i3} + G_{i4},\end{aligned}\quad (15)$$

where  $\Delta h_i = J_i \mathbf{M}_i^{-1} J_i [\mathbf{h}(\omega_i) \omega_i - \mathbf{h}(\hat{\omega}_i) \hat{\omega}_i] = J_i \mathbf{M}_i^{-1} J_i [\mathbf{h}(\omega_i) \mathbf{e}_{i2} - \mathbf{h}(\hat{\omega}_i) (\omega_i + \mathbf{e}_{i2})]$ ,  $l_i = |\mathbf{e}_{i1}|^{a_{i1}-1}$ . The stability condition of Eq. (15) can be obtained through the Routh-Hurwitz criterion as  $b_{i3} < 2b_{i1}b_{i2}$ .  $\square$

We construct a Lyapunov function  $V_i(e) = \Gamma_i^T P_i \Gamma_i$  with  $P_i$  as a positive definite symmetric matrix satisfying Eq. (16).

$$\begin{aligned}G_{i1}^T P_i + P_i G_{i1}^T &= -Q_1 \\ G_{i2}^T P_i + P_i G_{i2}^T &= -Q_2,\end{aligned}\quad (16)$$

where  $Q_1$  and  $Q_2$  are positive definite symmetric Hermitian matrices. The remaining proof procedure can be found in our existing research results [41]. Taking the derivation of the function  $V_i(e)$  and using Lemma 1 in the literature [41] can prove that the function  $V_i(e)$  converges in a fixed time, thereby obtaining stable convergence of estimation errors. For the solution of the fixed time, we can also refer to our existing research results [41], and we will not analyze them in detail here.

### 3.3. Self-triggered formation controller

The formation controller adopts a dual closed-loop structure, which can guarantee the convergence of the tracking error while satisfying the attitude, velocity and control input constraints. The UUV discrete model (17) can be obtained by using the forward Euler method with Eq. (1).

$$\begin{aligned}\eta_i(s+1|t_k^i) &= \eta_i(s|t_k^i) + J_i(s|t_k^i) \nu_i(s|t_k^i) T \\ &= f_1(\eta_i(s|t_k^i), \nu_i(s|t_k^i)) \\ \nu_i(s+1|t_k^i) &= [I - M_i^{-1}(C_i + D_i)T] \nu_i(s|t_k^i) + M_i^{-1} T \tau_i(s|t_k^i) \\ &= A_i \nu_i(s|t_k^i) + B_i \tau_i(s|t_k^i) = f_2(\nu_i(s|t_k^i), \tau_i(s|t_k^i))\end{aligned}\quad (17)$$

The outer-loop controller takes the amount of velocity change as input, which can control the velocity to change smoothly and suppress the effect caused by saturation control. The cost function of the outer-loop control problem of UUV<sub>i</sub> at moment  $t_k^i$  is

$$\begin{aligned}J_{i,N}^H(t_k^i) &= J_{i,N}^H(\eta_i(t_k^i), \eta_{-i}(t_k^i), \Delta \nu_i(t_k^i)) \\ &= \sum_{s=0}^{H_i-1} \frac{1}{\hbar} L_{i,1}[s|t_k^i] + \sum_{s=H_i}^{N-1} L_{i,1}[s|t_k^i] + E_{i,1}[N|t_k^i],\end{aligned}\quad (18)$$

where  $L_{i,1}[s|t_k^i] = \|\eta_{i0}(s|t_k^i)\|_{Q_1}^2 + \|\Delta \nu_i(s|t_k^i)\|_{R_1}^2 + \sum_{j \in \mathcal{N}_i} \|\eta_{ij}(s|t_k^i)\|_{Q_1'}^2$ , denotes the local generation value and  $E_{i,1}[N|t_k^i] = \|\eta_{i0}(N|t_k^i)\|_{P_1}^2$  is the terminal generation value.  $\Delta \nu_i(t_k^i) = \nu_i(t_k^i) - \nu_i(t_k^i - 1)$  is the velocity variation,  $\eta_{-i}(t_k^i) = (\eta_{i_1}(t_k^i), \dots, \eta_{i_{n_i}}(t_k^i))$  is the predicted state information of adjacent UUVs,  $i_{n_i} \in \mathcal{N}_i$ , and  $\Delta \nu_i(t_k^i)$  is the sequence of velocity variations.  $Q_1, Q_1', R_1$ , and  $P_1$  are the weighting matrices.  $H_i$  is the triggering interval, which is a single control period when  $H_i(t_k^i) = 1$ . The parameter  $\hbar \geq 1$  adjusts the control performance and the percentage of communication load, the larger the parameter  $\hbar$ , the more the system values communication load, and vice versa, the more it values the control performance. We achieve the self-triggered decision of intermittent communication by the outer-loop controller, whose control problem  $\mathcal{P}_{i,1}$  at the moment  $t_k^i$  is designed as

$$\begin{aligned}\min_{\Delta \nu_i(s|t_k^i)} &\left\{ J_{i,1}^H(\eta_i(t_k^i), \eta_{-i}(t_k^i), \Delta \nu_i(t_k^i)) \right. \\ &\quad \left. \text{such that } \eta_i(H_i|t_k^i) \in \Xi_{N-H_i}(\Omega_{i1}) \right\} \\ \text{s.t. } &\eta_i(s+1|t_k^i) = f_1(\eta_i(s|t_k^i), \nu_i(s|t_k^i)) \\ &\eta_i(0|t_k^i) = \eta_i(t_k^i) \\ &-\nu_{\max} \leq \nu_i(s|t_k^i) \leq \nu_{\max} \\ &-\nu_{\max} \leq \Delta \nu_i(s|t_k^i) \leq \Delta \nu_{\max} \\ &\Omega_{i1} = \{\eta_i(s|t_k^i) | \|\eta_{i0}(s|t_k^i)\|_{P_1}^2 \leq \sigma_{i1}\}, \sigma_{i1} \in \mathbb{R}^+\end{aligned}\quad (19)$$

where  $\nu_{\max}$  and  $\Delta\nu_{\max}$  are the constraint values for velocity and velocity increment,  $\Omega_{i1}$  is the terminal set and  $\Xi_{N-H_i}(\Omega_i)$  is the  $N - H_i$ -step robust set of  $UV_i$ . Define the optimal value of this control problem  $\mathcal{P}_{i,1}$  at the trigger interval  $H_i$  is  $V_{i,N}^{H_i}(\eta_i(t_k^i), \boldsymbol{\eta}_{-i}(t_k^i))$ . We design the self-triggering method to determine the optimal triggering interval  $H_i^*$  at moment  $t_k^i$ , which satisfies the relation  $t_{k+1}^i = t_k^i + H_i^*$ .  $H_i^*$  is calculated as follows:

$$H_i^* = \max\{H_i \in \mathbb{N}_{[1, \bar{H}] | V_{i,N}^{H_i}(t_k^i) \leq V_{i,N}^1(t_k^i)\}. \quad (20)$$

The velocity increment sequence  $\Delta\nu_i^*(t_k^i)$  corresponding to the optimal triggering interval  $H_i^*$  is taken as the optimal solution sequence to the control problem  $\mathcal{P}_{i,1}$  at the moment  $t_k^i$ , and then the optimal velocity sequence  $\nu_i^*(t_k^i)$  is obtained, which is taken as the expectation value of the inner-loop controller. The inner-loop controller will be solved  $H_i^*$  times within the triggering interval, and the cost function of its inner-loop control problem is:

$$\begin{aligned} J_{i,N}(t_k^i) &= J_{i,N}(\nu_i(t_k^i), \boldsymbol{\nu}_{-i}(t_k^i), \tau_i(t_k^i)) \\ &= \sum_{s=0}^{N-1} L_{i,2}[s|t_k^i] + E_{i,2}[N|t_k^i], \end{aligned} \quad (21)$$

where  $L_{i,2}[s|t_k^i] = \|\nu_{id}(s|t_k^i)\|_{Q_2}^2 + \|\tau_i(s|t_k^i)\|_{R_2}^2 + \sum_{j \in N_i} \|\nu_{ij}(s|t_k^i)\|_{Q_2'}^2$  denotes the local generation value and  $E_{i,2}[N|t_k^i] = \|\nu_{id}(N|t_k^i)\|_{P_2}^2$  is the terminal generation value.  $\tau_i(t_k^i)$  is the control input sequence, and  $\boldsymbol{\nu}_{-i}(t_k^i) = (\nu_{i_1}(t_k^i), \dots, \nu_{i_{n_i}}(t_k^i))$  is the predicted velocity information of adjacent UUVs.  $Q_2, Q_2', R_2$ , and  $P_2$  are the weighting matrices.

The inner-loop control problem  $\mathcal{P}_{i,2}$  at moment  $t_k^i$  is designed as follows:

$$\begin{aligned} \min_{\tau_i(s|t_k^i)} & \{J_{i,N}(\nu_i(t_k^i), \boldsymbol{\nu}_{-i}(t_k^i), \tau_i(t_k^i))\} \\ \text{s.t. } & \nu_i(s+1|t_k^i) = f_2(\nu_i(s|t_k^i), \tau_i(s|t_k^i)) \\ & \nu_i(0|t_k^i) = \nu_i(t_k^i) \\ & -\nu_{\max} \leq \nu_i(s|t_k^i) \leq \nu_{\max} \\ & -\tau_{\max} \leq \tau_i(s|t_k^i) \leq \tau_{\max} \\ & \Omega_{i2} = \{\nu_i(s|t_k^i) | \|\nu_{id}(s|t_k^i)\|_{P_2}^2 \leq \sigma_{i2}\}, \quad \sigma_{i2} \in \mathbb{R}^+ \end{aligned} \quad (22)$$

where  $\tau_{\max}$  is the constraint value of the control quantity and  $\Omega_{i2}$  is the terminal set. Solving problem  $\mathcal{P}_{i,2}$  yields the optimal control input  $\tau_i^*(t_k^i)$  for the nominal system, which is ideally unaffected by perturbations. In the motion of UUV, the control input  $\tau_i'$  that actually drives the UUV contains the control input  $\tau_i$  of the nominal system and the auxiliary control law  $\kappa_i e_i$ ,  $e_i = \nu_i - \nu_i'$ . Solving the Riccati equation yields the coefficient  $\kappa_i$ .

$$\begin{aligned} 2A_i^T P_i A_i + 2\kappa_i^T B_i^T P_i B_i \kappa_i - P_i + \rho + \kappa_i^T B_i^T P_i A_i \\ + A_i^T P_i B_i \kappa_i = -Q_2, \end{aligned} \quad (23)$$

where  $P_i = A_i A_i^T$ , the discrete model of an actual system with disturbances is as follows:

$$\nu_i'(s+1|t_k^i) = A_i \nu_i'(s|t_k^i) + B_i \tau_i'(s|t_k^i) + d_i(s|t_k^i). \quad (24)$$

Subtracting Eq. (17) and Eq. (24) yields an error model as

$$e_i(s+1|t_k^i) = A_i e_i(s|t_k^i) + B_i \kappa_i e_i(s|t_k^i) + d_i(s|t_k^i). \quad (25)$$

**Assumption 4.** The disturbance  $d_i$  satisfies the relationship  $d_i^T P_i d_i \leq e_i^T \rho e_i$  [45].

**Theorem 2.** When Assumption 4 holds, the velocity error  $e_i$  can gradually approach 0 through the control law  $\kappa_i e_i$ .

**Proof** By constructing a Lyapunov function  $\mathcal{F}_e = e_i^T P_i e_i$ , it can be shown that  $\mathcal{F}_e \geq 0$ . It is sufficient to show that  $\mathcal{F}_e$  is a monotonically decreasing function to conclude that the velocity error  $e_i$  tends to 0. Expanding and simplifying  $\Delta\mathcal{F}_e$  according to Assumption 4 and Lemma 1 [45] yields

$$\begin{aligned} \Delta\mathcal{F}_{e_i}(s+1|t_k^i) &= \mathcal{F}_{e_i}(s+1|t_k^i) - \mathcal{F}_{e_i}(s|t_k^i) \\ &\leq e_i^T (2A_i^T P_i A_i + 2\kappa_i^T B_i^T P_i B_i \kappa_i - P_i + \rho \\ &\quad + \kappa_i^T B_i^T P_i A_i + A_i^T P_i B_i \kappa_i) e_i \\ &\leq -e_i^T Q_2 e_i < 0. \end{aligned} \quad (26)$$

As a result, the system error  $e_i$  will gradually converge to 0, and the nonlinear auxiliary control law can suppress the effect of the perturbation.  $\square$

### 3.4. Theoretical analysis

The cost function of the outer-loop control problem can be rewritten as

$$J_{i,N}^{H_i}(t_k^i) = \sum_{s=0}^{H_i-1} \frac{1}{h} L_{i,1}[s|t_k^i] + V_{i,N-H_i}(\eta_i(t_k^i), \boldsymbol{\eta}_{-i}(t_k^i)), \quad (27)$$

$$V_{i,l}(\eta_i(t_k^i), \boldsymbol{\eta}_{-i}(t_k^i))$$

$$= \min_{\nu_i(s|t_k^i)} \left\{ \begin{aligned} & L_{i,1}(\eta_i(s|t_k^i), \boldsymbol{\eta}_{-i}(s|t_k^i), \nu_i(s|t_k^i)) \\ & + V_{i,l-1}(f_1(\eta_i(s|t_k^i), \nu_i(s|t_k^i)), \\ & \quad \times \boldsymbol{\eta}_{-i}(s+1|t_k^i)) \\ & \text{such that } \eta_i(s|t_k^i) \in \Xi_l(\Omega_{i1}), \\ & f(\eta_i(s|t_k^i), \nu_i(s|t_k^i)) \in \Xi_{l-1}(\Omega_{i1}) \end{aligned} \right\}. \quad (28)$$

**Assumption 5.** For  $UV_i$ , there exists a set of terminal constraints  $\Omega_{i1}$  with terminal controllers  $\mathcal{K}_i(\eta_i(t_k^i))$  [40],

and there exists a  $\mathcal{K}$ -function  $\sigma_i(\cdot)$  and a constant  $\alpha$ , such that

$$\begin{aligned} E_{i,1}(\eta_i(s+1|t_k^i)) - E_{i,1}(\eta_i(s|t_k^i)) \\ \leq L_{i,1}(\eta_i(s|t_k^i), \eta_{-i}(s|t_k^i), \mathcal{K}_i) + \varpi_i, \end{aligned} \quad (29)$$

where  $\varpi_i = \sigma_i(\|d_i(s)\|) + \alpha$ .

**Feasibility:** Based on the mathematical induction method to prove the feasibility of the formation controller, assuming that for  $UUV_i$ , its control problem  $\mathcal{P}_{i,1}$  has an optimal solution  $\nu_i^*(t_k^i) = (\nu_i^*(0|t_k^i), \dots, \nu_i^*(N-1|t_k^i))$  satisfying the constraints at moment  $t_k^i$ , the candidate velocity sequence  $\tilde{\nu}_i(t_{k+1}^i)$  at moment  $t_{k+1}^i$  can be constructed as follows:

$$\tilde{\nu}_i(s|t_{k+1}^i) = \begin{cases} \nu_i^*(s + H_i^*|t_k^i), & s \in \mathbb{N}_{[0, N-H_i^*-1]} \\ \mathcal{K}_i(\tilde{\eta}_i(s|t_k^i)), & s \in \mathbb{N}_{[N-H_i^*, N-1]} \end{cases}. \quad (30)$$

Its corresponding system state is  $\tilde{\eta}_i(s+1|t_{k+1}^i) = f_1(\tilde{\eta}_i(s|t_{k+1}^i), \tilde{\nu}_i(s|t_{k+1}^i))$ , where  $\tilde{\eta}_i(0|t_{k+1}^i) = \tilde{\eta}_i(t_k^i)$ , and it is known that  $\tilde{\nu}_i(t_{k+1}^i)$  satisfies the constraint by Assumption 5 and  $\nu_i^*(t_k^i)$ .

When  $s \in \mathbb{N}_{[0, N-H_i^*-1]}$ , the system states  $\tilde{\eta}_i(s|t_{k+1}^i)$  satisfy the constraints.

When  $s \in \mathbb{N}_{[N-H_i^*, N-1]}$ , the system states remain in the set of constraints through the terminal controller  $\mathcal{K}_i(\tilde{\eta}_i(s|t_{k+1}^i))$ . Therefore, the system states all satisfy the constraints, proving the feasibility of this controller.

**Stability:** We prove the stability of this formation controller by proving that the function  $V_{i,N}^{H_i^*}$  in the control problem  $\mathcal{P}_{i,1}$  is the ISpS function.

The local cost function  $L_{i,1}$  in Problem  $\mathcal{P}_{i,1}$  is quadratic form, and  $Q_1$ ,  $Q'_1$ , and  $R_1$  are positive definite matrices. It is known that there exists  $\mathcal{K}_\infty$ -function  $\alpha_L(\|\eta_i(t_k^i)\|) = \lambda(Q_1)\|\eta_i(t_k^i)\|^2$  such that  $L_{i,1} \geq \alpha_L(\|\eta_i(t_k^i)\|)$ , i.e.,  $V_{i,N}^{H_i^*} \geq \alpha_L(\|\eta_i(t_k^i)\|)$ .

Based on these assumptions, it can be concluded that

$$\begin{aligned} E_{i,1}(\eta_i(s+1|t_k^i)) - E_{i,1}(\eta_i(s|t_k^i)) \\ \leq -L_{i,1}(\eta_i(s|t_k^i), \eta_{-i}(s|t_k^i), \mathcal{K}_i(\eta_i(s|t_k^i))) + \varpi_i. \end{aligned} \quad (31)$$

Adding Eq. (31) from 0 to  $N-1$  yields

$$\begin{aligned} \sum_{s=0}^{N-1} L_{i,1}(\eta_i(s|t_k^i), \eta_{-i}(s|t_k^i), \mathcal{K}_i(\eta_i(s|t_k^i))) \\ + E_{i,1}(\eta_i(N|t_k^i)) \leq E_{i,1}(\eta_i(t_k^i)) + N\varpi_i, \end{aligned} \quad (32)$$

where  $E_{i,1}(\eta_i(t_k^i)) = E_{i,1}(\eta_i(0|t_k^i))$ ; when Eq. (20) holds,  $V_{i,N}^{H_i^*} \leq V_{i,N}^1$ .

Expression for the function  $E_{i,1}$  yields the existence of a  $\mathcal{K}_\infty$ -function  $\alpha_E(\|\eta_i(t_k^i)\|) = \bar{\lambda}(P_1)\|\eta_i(t_k^i)\|^2$  such that  $E_{i,1}(\eta_i(t_k^i)) \leq \alpha_E(\|\eta_i(t_k^i)\|)$ , i.e.,  $V_{i,N}^{H_i^*} \leq V_{i,N}^1 \leq V_{i,N} \leq \alpha_E(\|\eta_i(t_k^i)\|) + N\varpi_i$ .

From Eqs. (18) and (27), we obtain

$$\begin{aligned} J_{i,N-H_i^*}(\eta_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i), \nu_i^*(t_k^i)) \\ = J_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i), \nu_i^*(t_k^i)) \\ - \frac{1}{h} \sum_{s=0}^{H_i^*-1} L_{i,1}(\eta_i(s|t_k^i), \eta_{-i}(s|t_k^i), \nu_i^*(s|t_k^i)) \\ \leq J_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i), \nu_i^*(t_k^i)) - \frac{H_i^*}{h} \alpha_L(\|\eta_i(t_k^i)\|), \end{aligned} \quad (33)$$

where the optimal velocity sequence is  $\nu_{i,N-H_i^*}^*(t_k^i) = (\nu_i^*(H_i^*|t_k^i), \dots, \nu_i^*(N-1|t_k^i))$ .

The state  $\eta_i(t_{k+1}^i)$  of  $UUV_i$  can enter the terminal set under the action of a feasible solution, which follows from inequality (33) and Assumption 5:

$$\begin{aligned} J_{i,N}(\tilde{\eta}_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i), \tilde{\nu}_i(t_{k+1}^i)) \\ \leq J_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i), \nu_i^*(t_k^i)) - \frac{H_i^*}{h} \alpha_L(\|\eta_i(t_k^i)\|) \\ + H_i^* \varpi_i. \end{aligned} \quad (34)$$

When  $t_k^i - t_k^{ij} < \tau_k^{ij} \leq T_{\max} \cdot T$ , the predicted state information of the neighbors of  $UUV_i$  is  $\eta_{-i}(t_{k+1}^i) = (\eta_{-i}(H_i^*|t_k^i), \dots, \eta_{-i}(H_i^* + N-1|t_k^i))$ .

Through the proof procedure in [46], it can be concluded that the relationship  $V_{i,N}^1 \leq V_{i,N}$ .

$$\begin{aligned} V_{i,N}(\eta_i(t_k^i), \eta_{-i}(t_k^i)) &= J_{i,N}(\eta_i(t_k^i), \eta_{-i}(t_k^i), \nu_i^*(t_k^i)) \\ &= \sum_{s=0}^{N-1} L_{i,1}(\eta_i(s|t_k^i), \eta_{-i}(s|t_k^i), \nu_i^*(s|t_k^i)) \\ &\quad + E_{i,1}(\eta_i(N|t_k^i)). \end{aligned} \quad (35)$$

This is obtained through the above inequality and Eq. (34):

$$\begin{aligned} V_{i,N}^{H_i^*}(\eta_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i)) \\ \leq V_{i,N}^1(\eta_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i)) \\ \leq V_{i,N}(\eta_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i)) \\ = J_{i,N}(\eta_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i), \nu_i^*(t_{k+1}^i)) \\ \leq J_{i,N}(\eta_i(t_{k+1}^i), \eta_{-i}(t_{k+1}^i), \tilde{\nu}_i(t_{k+1}^i)) \\ \leq J_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i), \nu_i^*(t_k^i)) - \frac{H_i^*}{h} \alpha_L(\|\eta_i(t_k^i)\|) \\ + H_i^* \varpi_i \\ \leq V_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i)) - \frac{H_i^*}{h} \alpha_L(\|\eta_i(t_k^i)\|) + H_i^* \varpi_i. \end{aligned} \quad (36)$$

When  $0 \leq \tau_k^{ij} \leq t_k^i - t_k^{ij}$ , new neighbor information  $\eta_{-i}(t_{k+1}^i) = (\eta_{-i}(0|t_{k+1}^i), \dots, \eta_{-i}(N-1|t_{k+1}^i))$  will be used. Due to  $\eta_i(\cdot|t_k^i) \in \mathbb{X}_i$ , there exists a constant  $\delta_i$  such that  $\|\eta_i(\cdot|t_k^i)\| \leq \delta_i$ .

From the trigonometric inequality and the state constraints, the following relationship can be obtained:

When  $s \in [0, N - H_i^* - 1]$ , we obtain

$$\begin{aligned} & \|\eta_i(s|t_{k+1}^i) - \eta_j(s|t_{k+1}^i) - \Delta_{ij}\| \\ & \leq \|\eta_i(s|t_{k+1}^i) - \eta_j(H_i^* + s|t_k^i) - \Delta_{ij}\| \\ & \quad + \|\eta_j(s|t_{k+1}^i) - \eta_j(H_i^* + s|t_k^i)\| \\ & \leq \|\eta_i(s|t_{k+1}^i) - \eta_j(H_i^* + s|t_k^i) - \Delta_{ij}\| + 2\delta_i. \end{aligned} \quad (37)$$

Next, we consider  $s \in [N - H_i^*, N - 1]$ .

$$\begin{aligned} & \|\eta_i(s|t_{k+1}^i) - \eta_j(s|t_{k+1}^i) - \Delta_{ij}\| \\ & \leq \|\eta_i(s|t_{k+1}^i) - \Delta_{ij}\| + \|\eta_j(s|t_{k+1}^i)\| \leq \|\eta_i(s|t_{k+1}^i) \\ & \quad - \Delta_{ij}\| + \delta_i. \end{aligned} \quad (38)$$

The coupling term in the local cost function can be simplified by Eqs. (37) and (38):

$$\begin{aligned} & \sum_{s=0}^{N-1} \left( \sum_{j \in \mathcal{N}_i} \|\eta_{ij}(s|t_{k+1}^i)\|_{Q_1'}^2 \right) \\ & = \sum_{s=0}^{N-1} \left( \sum_{j \in \mathcal{N}_i} \|\eta_i(s|t_{k+1}^i) - \eta_j(s|t_{k+1}^i) - \Delta_{ij}\|_{Q_1'}^2 \right) \\ & \leq \sum_{s=0}^{N-H_i^*-1} \sum_{j \in \mathcal{N}_i} (\|\eta_i(s|t_{k+1}^i) - \eta_j(H_i^* + s|t_k^i) - \Delta_{ij}\|_{Q_1'} \\ & \quad + 2\delta_i)^2 + \sum_{s=N-H_i^*}^{N-1} \sum_{j \in \mathcal{N}_i} (\|\eta_i(s|t_{k+1}^i) - \Delta_{ij}\|_{Q_1'} + \delta_i)^2 \\ & \leq \sum_{s=0}^{N-1} \left( \sum_{j \in \mathcal{N}_i} \|\eta_i(s|t_{k+1}^i) - \eta_j(H_i^* + s|t_k^i) - \Delta_{ij}\|_{Q_1'}^2 \right) \\ & \quad + H_i^* \bar{\lambda}(Q_1') \sum_{j \in \mathcal{N}_i} (3\delta_i^2 + 2\delta_i \Delta_{ij}) \\ & \quad + (N - H_i^*) \bar{\lambda}(Q_1') \sum_{j \in \mathcal{N}_i} (12\delta_i^2 + 4\delta_i \Delta_{ij}). \end{aligned} \quad (39)$$

Through Eq. (34) and the above inequality relation, we can obtain

$$\begin{aligned} & V_{i,N}^{H_i^*}(\eta_i(t_{k+1}^i), \hat{\eta}_{-i}(t_{k+1}^i)) \leq V_{i,N}^1(\eta_i(t_{k+1}^i), \hat{\eta}_{-i}(t_{k+1}^i)) \\ & \leq V_{i,N}(\eta_i(t_{k+1}^i), \hat{\eta}_{-i}(t_{k+1}^i)) = J_{i,N}(\eta_i(t_{k+1}^i), \\ & \quad \times \hat{\eta}_{-i}(t_{k+1}^i), \nu_i^*(t_{k+1}^i)) \\ & \leq J_{i,N}(\eta_i(t_{k+1}^i), \hat{\eta}_{-i}(t_{k+1}^i), \tilde{\nu}_i(t_{k+1}^i)) \leq J_{i,N}(\eta_i(t_{k+1}^i), \\ & \quad \times \eta_{-i}(t_{k+1}^i), \tilde{\nu}_i(t_{k+1}^i)) + (N - H_i^*) \bar{\lambda}(Q_1') \sum_{j \in \mathcal{N}_i} \\ & \quad \times (12\delta_i^2 + 4\delta_i \Delta_{ij}) + H_i^* \bar{\lambda}(Q_1') \sum_{j \in \mathcal{N}_i} (3\delta_i^2 + 2\delta_i \Delta_{ij}) \\ & \leq J_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i), \nu_i^*(t_k^i)) - \frac{H_i^*}{\hbar} \alpha_L(\|\eta_i(t_k^i)\|) + \Theta_i \\ & \leq V_{i,N}^{H_i^*}(\eta_i(t_k^i), \eta_{-i}(t_k^i)) - \frac{H_i^*}{\hbar} \alpha_L(\|\eta_i(t_k^i)\|) + \Theta_i, \end{aligned} \quad (40)$$

where  $\Theta_i = (N - H_i^*) \bar{\lambda}(Q_1') \sum_{j \in \mathcal{N}_i} (12\delta_i^2 + 4\delta_i \Delta_{ij}) + H_i^* \bar{\lambda}(Q_1') \sum_{j \in \mathcal{N}_i} (3\delta_i^2 + 2\delta_i \Delta_{ij}) + H_i^* \varpi_i$ .

Based on Lemma 1, we prove in detail that the function  $V_{i,N}^{H_i^*}$  is an ISpS Lyapunov function, and the same applies for the inner-loop controller proof procedure.

#### 4. Simulation Example

This section verifies the effectiveness and control performance of the proposed self-triggered formation controller through numerical experiments in two scenarios. The formation system consists of a virtual leader UAV and multiple following UAVs. Figure 4 shows the structure diagram of the target formation, with each following UAV maintaining a distance of 5 m from  $UUV_0$  located at the center of the formation. Figure 5 shows the communication network diagram of the formation system, which is a directed spanning tree with  $UUV_0$  as the root node. The detailed physical parameters of the UAV system selected in the experiment can be found in the literature [35].

The parameters in the numerical experiments are chosen as:  $T = 0.2$  s,  $N = 15T$ ,  $H_{\max} = 5$ , and  $T_{\max} = 5$ . The parameters of the formation controller are  $Q_1 = \text{diag}\{1, 3, 4, 6, 5\}$ ,

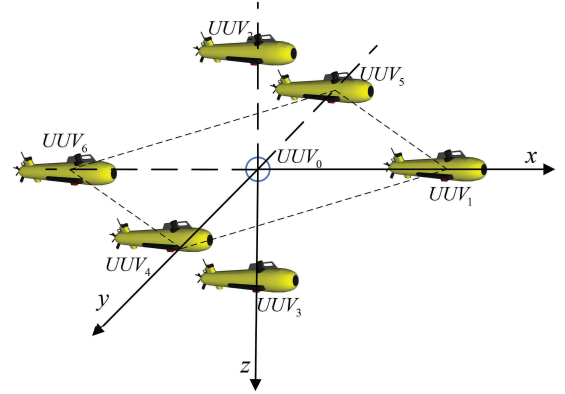


Fig. 4. Target formation structure chart.

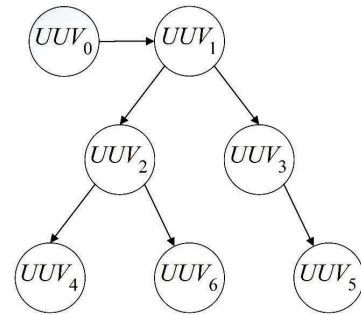


Fig. 5. Formation communication schematic.

$Q'_1 = \text{diag}\{1, 1, 1, 1, 1\}$ ,  $R_1 = \text{diag}\{8, 8, 8, 2, 2\}$ ,  $Q_2 = \text{diag}\{1, 1, 3, 3, 8\}$ ,  $Q'_2 = \text{diag}\{1, 1, 1, 1, 1\}$ , and  $R_2 = \text{diag}\{10^{-3}, 10^{-3}, 10^{-3}, 10^{-3}, 10^{-3}\}$ . The velocity constraints are  $\nu_{\max} = [2.5; 0.6; 0.6; \pi/18; \pi/9]$  and  $\Delta \nu_{\max} = [0.1; 0.08; 0.08; \pi/36; \pi/18]$ , and the control input constraints are  $\tau_{\max} = [350; 350; 350; 150; 150]$ . The parameters of the extended state observer are  $a_{i1} = 0.85$ ,  $b_{i1} = 0.4$ ,  $b_{i2} = 0.2$ , and  $b_{i3} = 0.001$ . The communication delay  $t_k^{ij}$  is an integer multiple of the control period  $T$  and varies randomly.

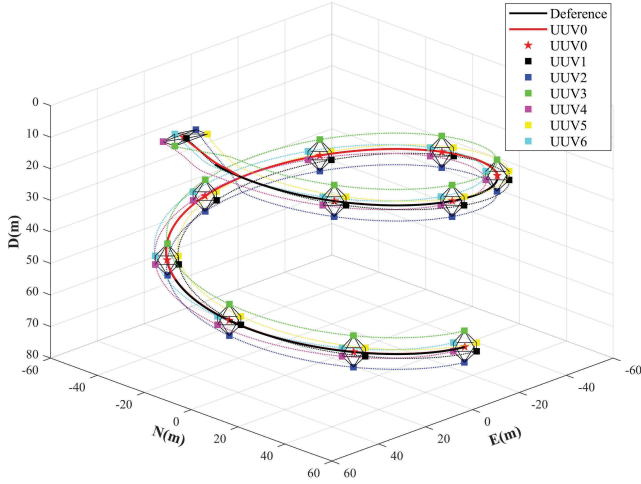


Fig. 6. UUV formation tracking trajectory.

The initial position of each UUV in the formation is on the water surface of  $[-20, -5] \times [40, 60]$ , and the initial velocity is 0. The formation system needs to form and maintain the target formation during the movement, and the position deviations between each following UUV and the leader UUV are:  $\Delta_1 = [5 \ 0 \ 0 \ 0 \ 0]$ ,  $\Delta_2 = [0 \ 0 \ 5 \ 0 \ 0]$ ,  $\Delta_3 = [0 \ 0 \ -5 \ 0 \ 0]$ ,  $\Delta_4 = [0 \ 5 \ 0 \ 0 \ 0]$ ,  $\Delta_5 = [0 \ -5 \ 0 \ 0 \ 0]$ , and  $\Delta_6 = [-5 \ 0 \ 0 \ 0 \ 0]$ .

**Simulation 1:** Check the effectiveness and convergence performance of the proposed formation controller under the influence of time-varying communication delays. Each following UUV in the formation tracks the reference trajectory by maintaining the desired position deviation from the virtual UUV, and we choose the spiral diving curve as the reference trajectory of the virtual UUV with the following expression:

$$\begin{cases} x_0 = 50 \sin(0.01t) \\ y_0 = 50 \cos(0.01t), \quad t \leq 800. \\ z_0 = 5 + 0.075t \end{cases} \quad (41)$$

Figures 6 and 7 depict the trajectory tracking effect and the position attitude change curve of the UUV formation system, respectively. It can be intuitively seen from the figures that the UUV formation starts from the water surface, forms the expected formation and tracks the reference

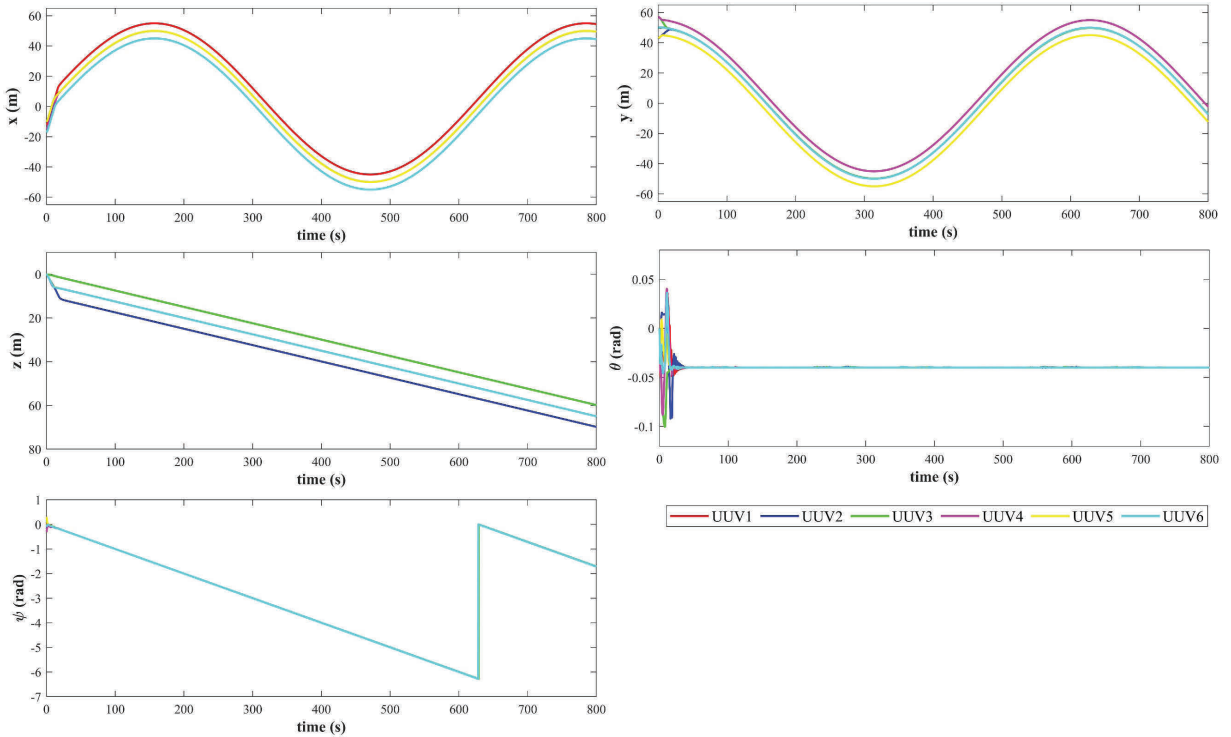


Fig. 7. UUV position and attitude change curve.

trajectory at 30 s, which indicates that the formation controller can effectively achieve the formation tracking task, and the attitude angle quickly converges to a stable state. Figures 8 and 9 show the velocity and control input plots over time, respectively, from which it can be seen that the formation system achieves stable convergence at approximately 30 s, and the velocity and control inputs remain stable after 30 s. Figure 10 depicts the tracking error variation curves for each UUV, and the tracking error rapidly converges within 30 s, and the error remains within 0.1 m. The experimental results verify that the self-triggered formation controller can effectively suppress the effect of communication delay on formation control, and quickly and accurately form the desired formation and track the reference trajectory.

**Simulation 2:** To better demonstrate the control performance of this self-triggered formation controller, the model parameter perturbation and bounded disturbance problems are considered in the second set of simulation experiments, where there is 15% uncertainty in the model parameters of the UUVs, and the bounded disturbance  $d_i$  is selected as

$$d_i(t) = \begin{bmatrix} 0.25u_i^3 + 0.25 \sin(0.05t) \\ 0.3v_i^3 + 0.2 \cos(0.05t) \\ 0.15w_i^3 + 0.1 \sin(0.05t + \pi/4) \end{bmatrix}. \quad (42)$$

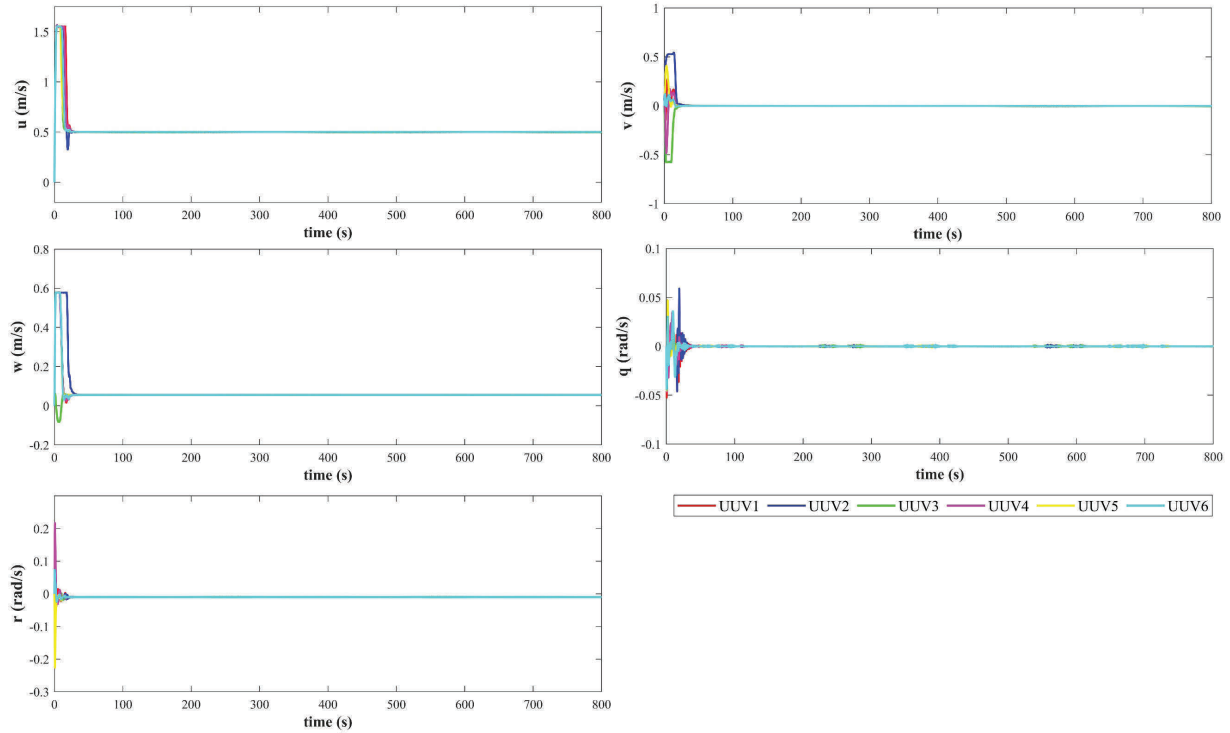


Fig. 8. UUV velocity change curve.

This group of simulation experiments simulates the UUV system to perform ocean missions, which requires the UUV to track the U-shaped route after completing the spiral diving. The reference trajectory of the virtual UUV is shown in Eqs. (43)–(46):

(1) Spiral diving curve

$$\begin{cases} x_0 = 50 \sin(0.01t) \\ y_0 = 50 \cos(0.01t), \quad t \leq 250\pi. \\ z_0 = 5 + 0.075t \end{cases} \quad (43)$$

(2) U-shaped curve

$$\begin{cases} x_0 = 50 \\ y_0 = -0.5025(t - 250\pi), \quad 250\pi < t \leq 250\pi + 200, \\ z_0 = 5 + 18.75\pi \end{cases} \quad (44)$$

$$\begin{cases} x_0 = 50 \sin(0.01t - 2\pi - 2) \\ y_0 = -100.5 + 50 \cos(0.01t - 2\pi - 2), \\ \quad 250\pi + 200 < t \leq 350\pi + 200 \\ z_0 = 5 + 18.75\pi \end{cases}, \quad (45)$$

$$\begin{cases} x_0 = -50 \\ y_0 = -100.5 + 0.5025(t - 350\pi - 200), \\ \quad 350\pi + 200 < t \leq 350\pi + 400 \\ z_0 = 5 + 18.75\pi \end{cases}. \quad (46)$$

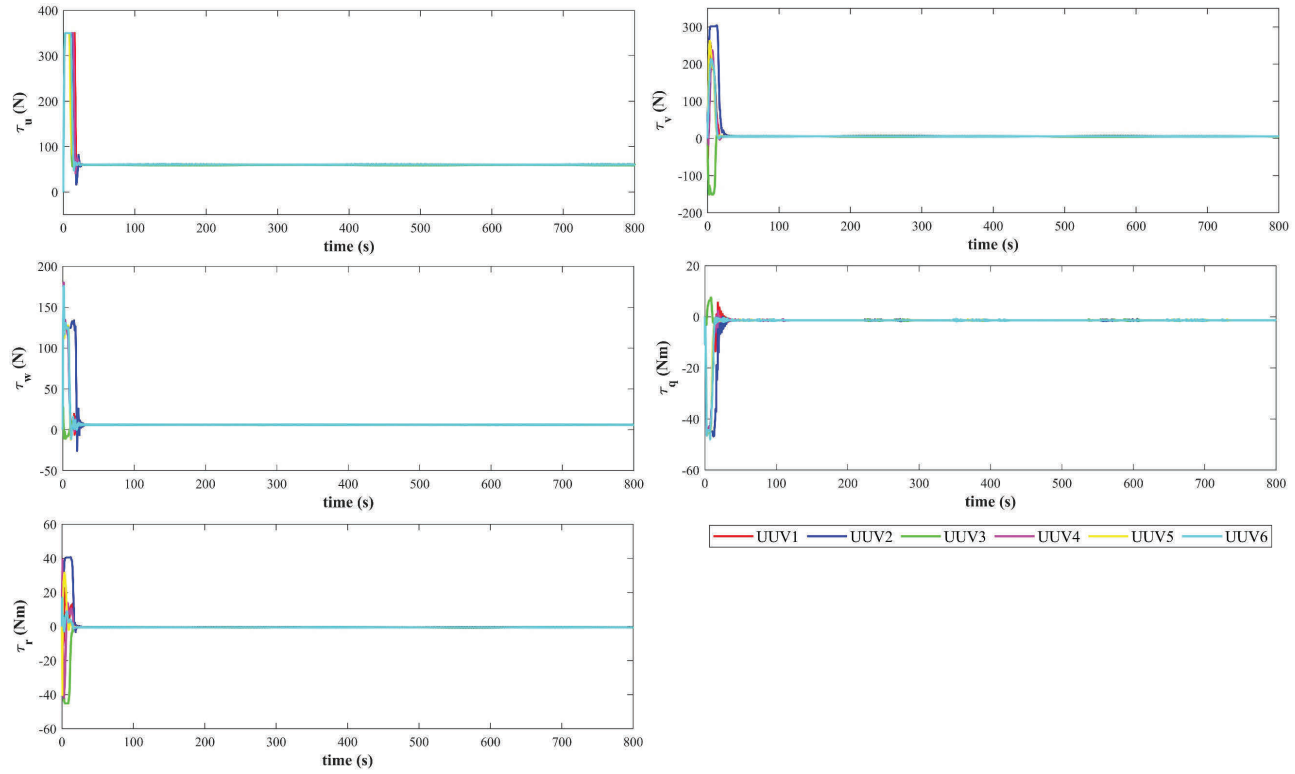


Fig. 9. UUV control input change curve.

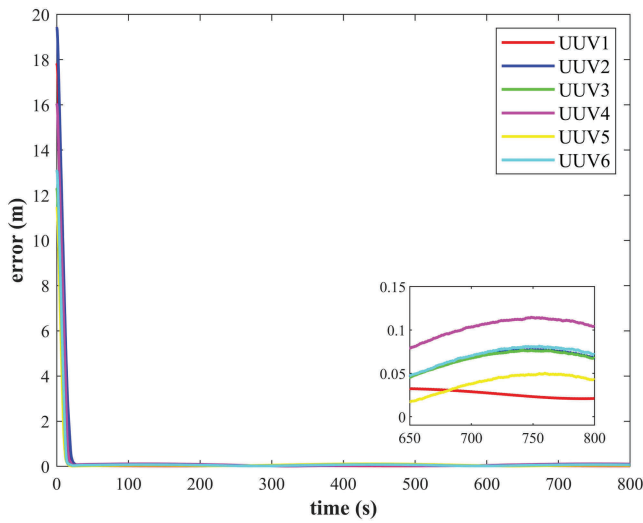


Fig. 10. Time-variation curve of formation tracking error.

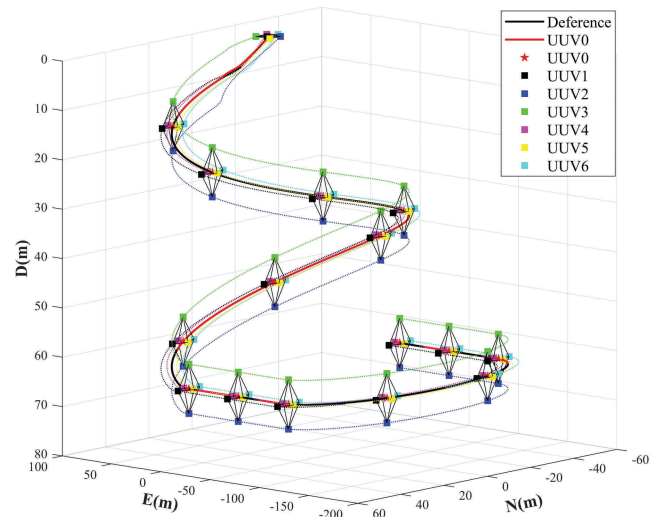


Fig. 11. UUV formation tracking trajectory.

The experimental results of the self-triggered formation controller in the complex environment are shown in Figs. 11–16. Figures 11 and 12 depict the trajectory and position attitude curve of the formation system over time. The figures demonstrate that the UUV system can quickly form a target

formation and achieve smooth transitions between different routes, and accurately track the spiral diving curve and U-shaped route. Figures 13 and 14 show the velocity and control input over time graphs. The inner and outer loop control structure enhances the controllability of the system over the

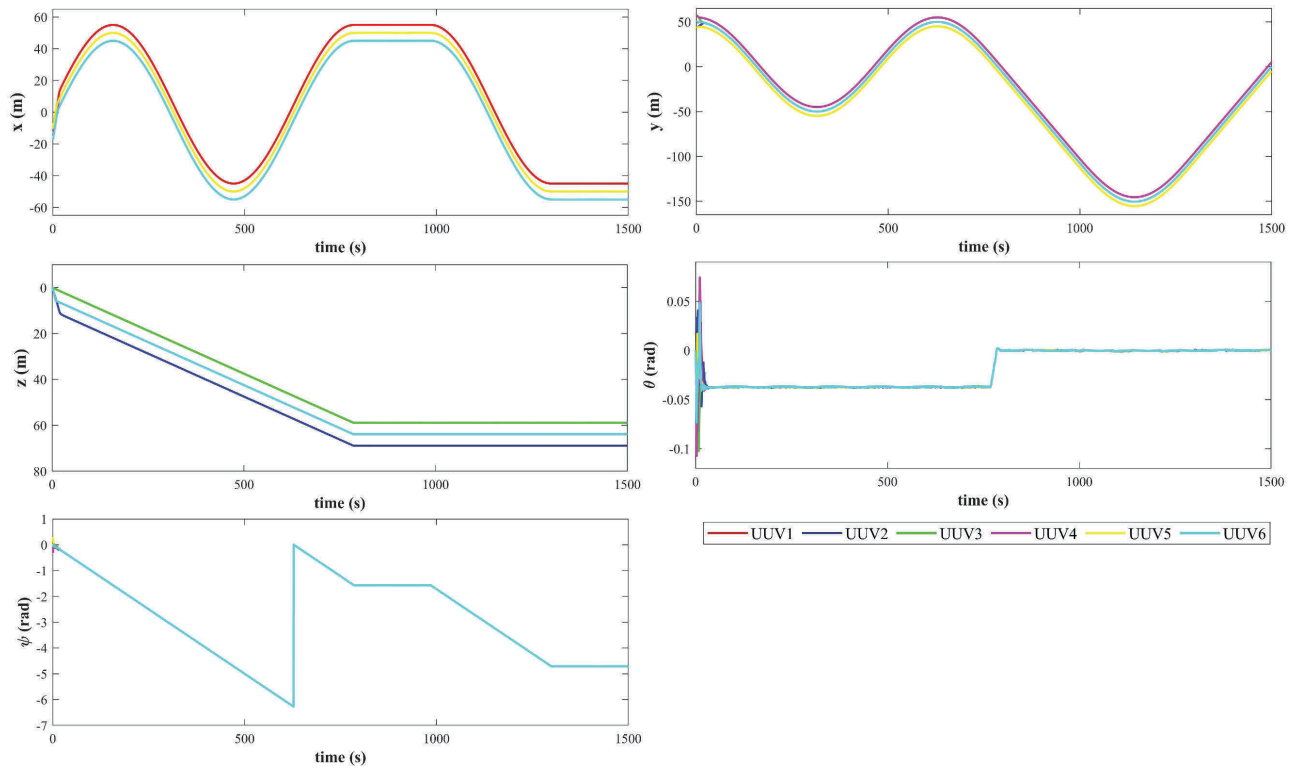


Fig. 12. UUV position and attitude change curve.

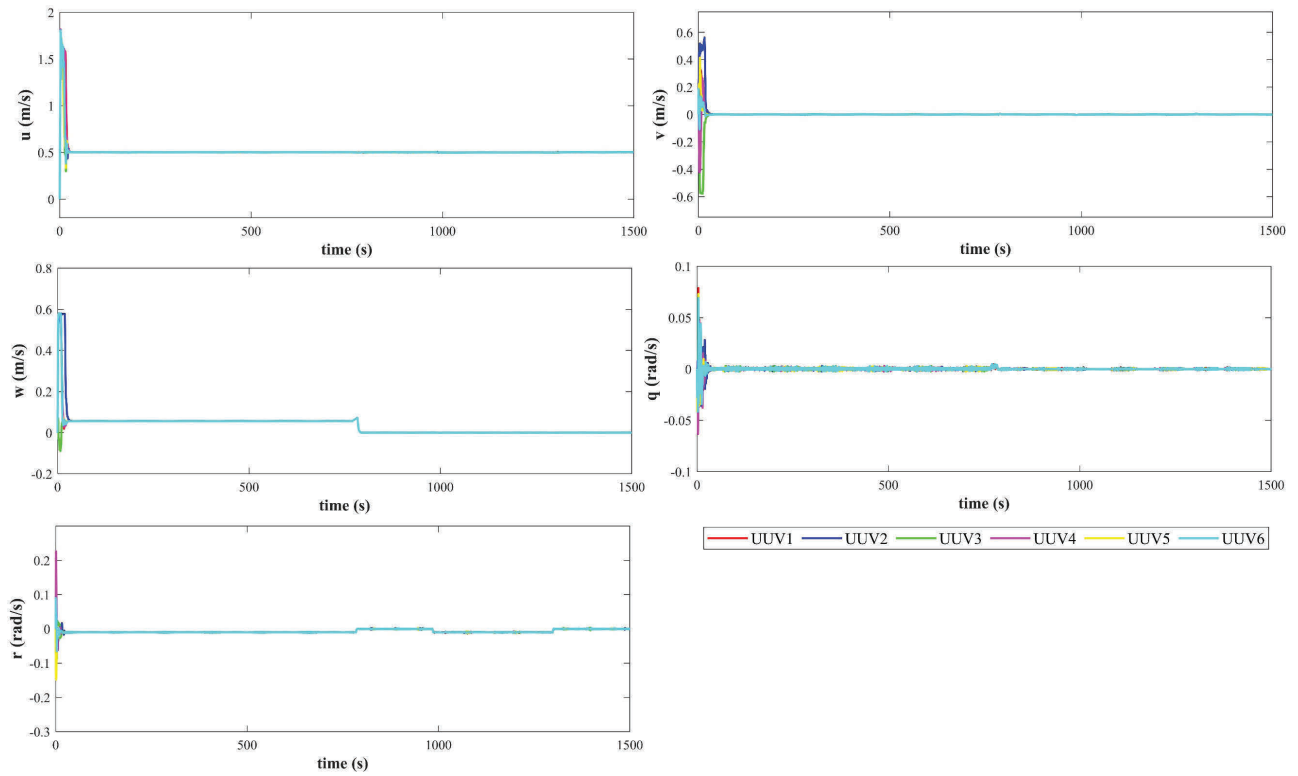


Fig. 13. UUV velocity change curve.

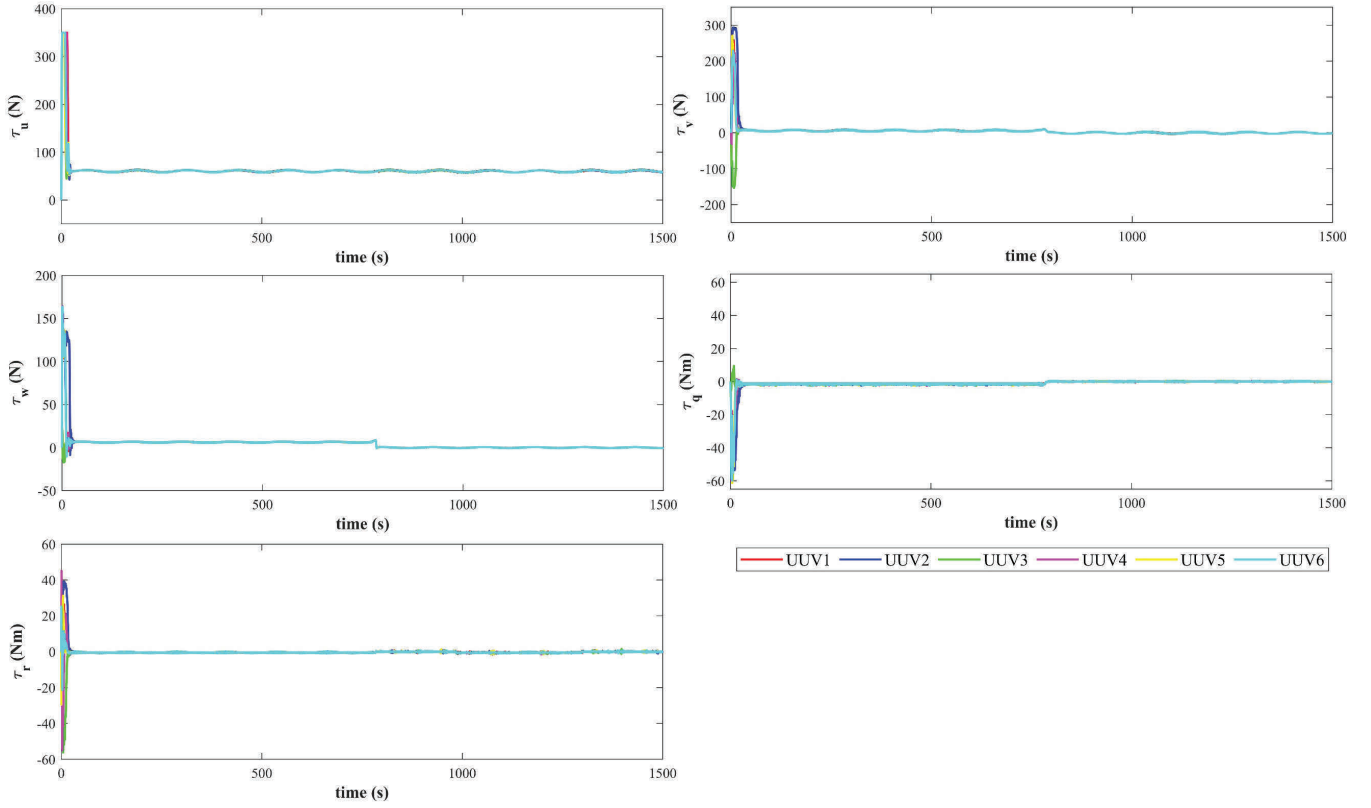


Fig. 14. UUV control input change curve.

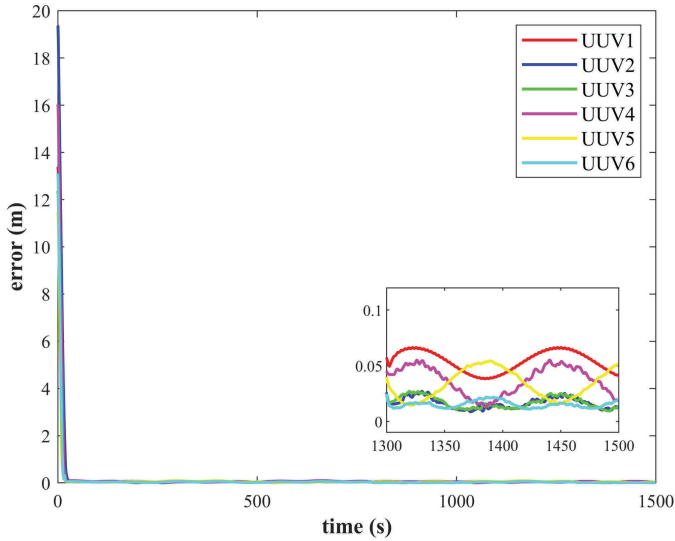


Fig. 15. Time-variation curve of formation tracking error.

velocity, and the auxiliary control law in the inner loop controller counteracts the influence of the sea current on the velocity by compensating for the control input, so the velocity can be kept constant. However, the control input may exhibit small fluctuations due to disturbances. Figure 15 shows the formation tracking error curve, which roughly shows that the tracking error converges within 30 s, and the error of all UUVs can be maintained within 0.07 m. Figure 16 shows the triggering situation of each UUV in 60–78 s, during which the formation system gradually converges to a stable state and the communication interval between UUVs will stabilize at 5 control cycles. Table 1 lists the number of communication triggers of each UUV during the motion process. If each control cycle is communicated, each UUV needs to communicate 7500 times, while the proposed self-triggered controller needs only approximately 1650 times. This can greatly relieve the communication load of the formation system, reduce the information redundancy, and improve the communication efficiency.

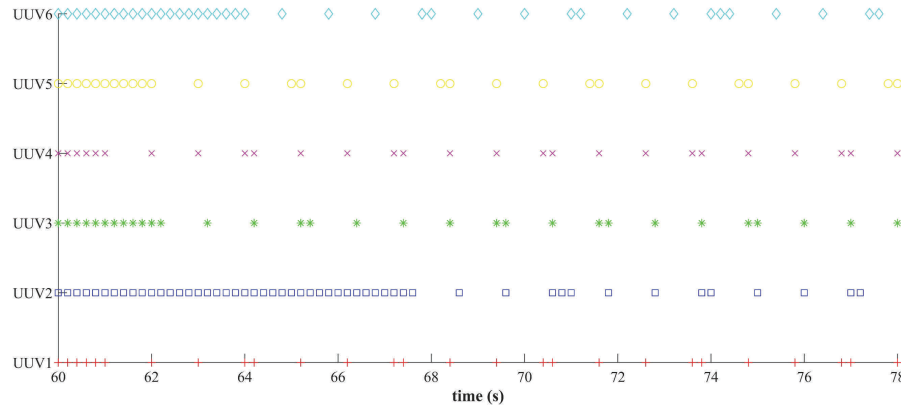


Fig. 16. 60–78 s for each UUV trigger moment.

Table 1. Communication frequency of each UUV.

|                  | UUV1 | UUV2 | UUV3 | UUV4 | UUV5 | UUV6 |
|------------------|------|------|------|------|------|------|
| Periodic control | 7500 | 7500 | 7500 | 7500 | 7500 | 7500 |
| Self-triggered   | 1674 | 1638 | 1650 | 1623 | 1642 | 1645 |

## 5. Conclusion

In this paper, the formation control problem of a multi-UUV formation system under bounded communication delays under the influence of model parameter perturbations and external disturbances is investigated, and a self-triggered distributed robust model prediction controller with a dual closed-loop structure is proposed. The RMPC and FESO in the inner-loop controller work together to improve the stability of the system under model parameter perturbations and external disturbances. Based on the self-triggered mechanism, asynchronous communication between individual UUVs is realized, which reduces the communication load. Finally, the closed-loop stability and feasibility of this self-triggered formation controller are rigorously analyzed theoretically. Simulation experiments verify the effectiveness of this controller.

Considering the complexity and danger of the UUV working environment, obstacle avoidance during navigation is an unavoidable challenge, and the cooperative obstacle avoidance of UUV formation system and mission replanning after a failure of a formation member will be studied in future work.

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